MANAGING "APPROACH CLIMB LIMIT" BEFORE TAKEOFF

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Let's say you are taking off from Lahore to land at Quetta (high elevation). First, as a routine you will check Box A, B and C to determine the max TOGW for takeoff from Lahore.

F15 :	: 17/03/2016	Į.	LLAMA IQBAL INTR	<mark>.</mark>		: :	LHE 36R	
T.O.R.A A.S.D.A T.O.D.A SLOPE	ION= 712.0 (FT) A. = 3360.0 (M) A. = 3482.0 (M) A. = 3665.0 (M) = 0.05 (%) WIDTH ASSUMED	0-DRY CHECK 1-STRUCTURE	TION CODES 5-TYRE SPEED 6-BRAKE ENERGY 7-RWY 2 ENGINES 8-FINAL T.O. 9-VMC	: : : :	ATR72-500 ATR72-500 V2/VS OPTIMIZED AIR COND. ON NORMAL CONDITIONS WITHOUT REVERSE	,	FAR V1/VR OPTI	
	: TOW(KG) COI : V1 VR V2(IAS : DTOW / I : DV1 DVR DV2 / DV1	KT) TOW I	NH=1013.25(HPA)	:	DRY RUNWAY SCREEN HEIGHT 35 F	T		
(DC) :			WIND (
:	-15	: -10 -:) 	: 10 :	: :	20	
36.0 :	: 21252 6-6 : 89 107 111 : -166 / +167 : +0 -1 -1/ +0 +0 +0	: -326 /	.18 : 105 11 +49 : +0 /	+0	•	-0 :	105 115	2-1 119 +0 3 -4

Let's say the temperature was 36 and you were box A limited to 22,800. This means you can takeoff with your maximum TOGW capability. But since you are landing at a high elevation airport with terrain all around, you will have to see if in case of a go around with single engine, your aircraft performance is sufficient to clear all the obstacles. High elevation means air is less dense so performance will be affected. So you can't afford to be very heavy. That's why you will check your approach climb limit before taking off (else it will be too late because you can't dump weight in air).

Assuming the expected temperature at Quetta is 36, your approach climb limit will be 19,769.

If you are above this weight you won't meet the required legal performance. So in addition to box A, B and C you calculate another limit. You may call it box D. Box D will be (as

:LDG CHART 21/04/2016	QUETTA		:	UET	31R	:
ELEVATION= 5267.0 (FT) :L.D.A. = 3658.0 (M) 0-V :SLOPE = 1.01(%) 1-S : 2-F	WET CHECK STRUCTURE CRUNWAY	<mark>3-approach clim</mark> 4-landing clime	B : ATR : ATR : APP	72-500 ROACH)	: : : : :
: : LANDING F30 : : : APPROACH F15 : : CAT I :: : OAT :DRY OR WET RWY	QNH = : : LANI :	= 1013.25(HPA) DING WEIGHT (KG CODE	-: :): : DRY (: : : OR WEI	LANDING APPROAC CAT I	F30 : H F15 : :
: (DC) :: : : WIND (F	KT) 0 : 10 :	:: 20 :: -15 :	WIN:	O (KT)	10 :	20 :
: 36.0 : 19769: 19769: 1976 : : 3 : 3 : 3	69: 19769: <mark>:</mark>	<mark>19769:</mark> : 19769:	19769:	19769:	19769:	19769:

given in SOP) Expected Approach Climb Limit Weight (determined from FOS, as shown above) + Trip fuel (from Lahore to Quetta) + Taxi fuel.

Your max TOGW will be lower of Box A, B, C and D. This example clearly shows that landing at Quetta when temperature is 36, you need to be at 19,769 Kgs. That means you can be above this weight at takeoff provided you lose weight during flight to be 19769 at landing. The only weight you can lose in flight is of fuel (trip and taxi) by consuming it. If in this case your trip fuel was 900 Kg and taxi 91 Kg, then your box D will be 19769 + 900 + 91 = 20,760. Much lower than 22,800.

In other words, if you plan a takeoff at 22800 then you will burn only 991 Kgs during flight (taxi + trip), and your weight at landing will be 21,809 Kgs which is higher than your approach climb limit of 19,769. In case of a single engine go around during approach, you won't meet the required performance.

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